

# GI420 Serials

## High Precision MEMS Integrated Navigation System

# **Technical Manual**



#### **GI420** <u>High Precision MEMS Integrated Nav</u>igation System



#### Introduction

BW-GI420 is a high-precision MEMS sensor with high reliability and stability. It also has the measurement accuracy of the medium and low precision fiber optic gyroscope and the excellent environmental characteristics of the MEMS system. It is coupled with high precision GPS system to provide high-precision position information.

BW - GI420 carry the newest integrated navigation sensor fusion algorithm engine. And this optimal design is developed for solving the multipath interference. It can meet the requirements of long time, high precision and high reliability navigation application in the complex environment of city and field. Our product has good expansibility, BW-GI420 can be added an external GNSS/ odometer/DVL/barometric altitude meter and other many kinds of sensors. Because of this, we can use the multi-sensor data fusion technology to combine the info of inertial measurement, satellite navigation, odometer and other senors' data to improve the adaptability and diversity of the system.

BW - GI420 integrated navigation system use tight coupling technique to combine the high precision, professional level, double antenna, multi-channel single frequency carrier phase and pseudo - range GPS receiver and high precision MEMS inertial measurement unit, so it has smaller volume, lighter weight, higher performance, lower cost and other advantages.

#### **Applications**

- Antenna stabilization platform, drone flight control
- Navigation and control of AGV
- Attitude measurement and orientation reference



## Specifications

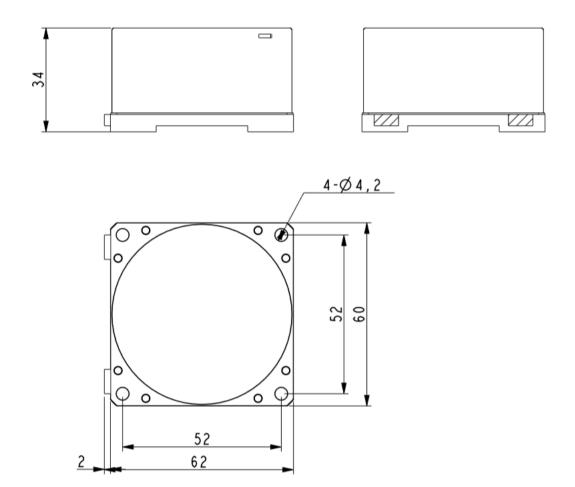
#### Performance Specifications

Attitude accuracy	(2m baseline)	0.1 deg	
	Satellite signal loss	1 deg /5min	
	Rolling attitude accuracy	≤0.2 deg	
	Pitch attitude accuracy	≤0.2 deg	
	Speed accuracy	0.1m/s	
Combined external GPS	Single point positioning accuracy	DGPS accuracy 0.5m /RTK accuracy 2cm	
	Directional accuracy	0.1deg, (2m baseline)	
	Azimuth	±180 deg	
	Roll angle	±90 deg	
Measuring range	Pitch angle	±90 deg	
	Angular velocity	±300 deg/s	
	Acceleration	±4 g	
	Power supply	9~36 VDC, ≤3 W @ 24VDC	
Interface characteristics	Electrical interface	RS422	
characteristics	Data update rate	100 Hz @ 115,200 baud rate	
	Operating temperature	-40°C ~ +60°C	
Environment	Storage temperature	-55°C ~ +85°C	
Physical characteristics	Size	60×62× 34 mm	
	Weight	≤100 g	
	Connector	J30J-15TJL	



#### **Product Size**

Product Size: L60×W62×H34 (mm)



Note: Unfilled dimensional tolerances are performed in accordance with GB/T1804-2000 Class C.

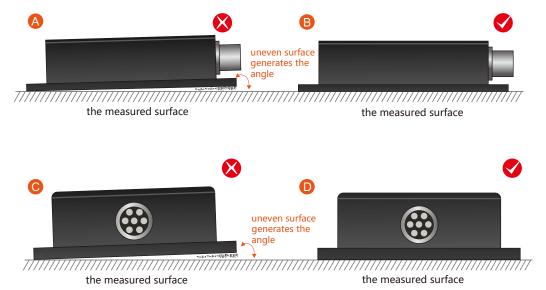


#### **Product installation**

The correct installation method can avoid measurement error. The inertial measurement unit produces an absolutely smooth surface by grinding on the underside and measuring surface for installation .The following points should be made when installing the sensor:



First of all, to ensure that the sensor mounting surface and the measured surface completely close, the measured surface should be as horizontal as possible, so that we can reduce installation error. Secondly, when the product is installed, the bottom surface of the IMU and the body should be coincident, and it is gently pushed to make the measurement surface of the IMU coincide with that of the machine body, so that both surfaces are tightly attached, and there is no angle in Fig.A and Fig.C.The correct installation is shown in Fig.B and Fig.D.



Finally, after the product is tightly attached, screws should be used to fix the IMU to ensure tight fixation, smooth conduct and stable rotation. Measurement errors caused by acceleration and vibration should be voided. Bear in mind that the screw only plays a fixed role and doe not have a positioning function. The screw hole of the IMU is processed into an elliptical shape for easy adjustment.



#### **Electrical connections**

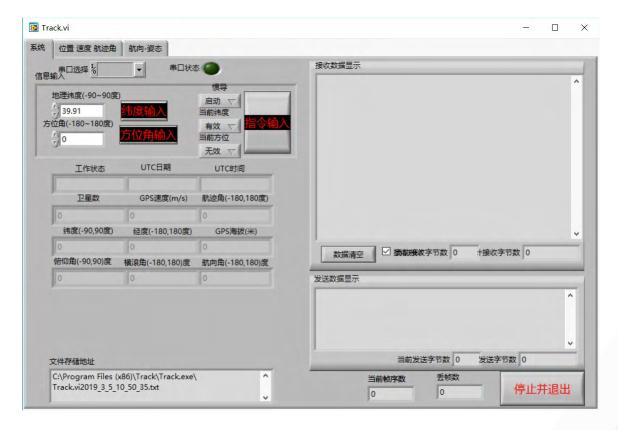
Data cable RS422 interface definition, DB9 female:

DB9 connector pin	DB9 connector pin Interface definition Remarks		
1	R+	Receive +	
2	R-	Receive -	
3	T+	Transmit +	
4	T-	Transmit -	

### **Debugging software**

#### Software usage steps:

- Ensure that the inertial navigation is absolutely static, correctly connect the serial port hardware of the integrated navigation, and connect the power supply.
- ② Select the computer serial port and baud rate and click to connect to the serial port.
- ③ Enter the correct geographic latitude, click Inertial Navigation Start→Command Input, the working status on the screen shows static alignment, and the working state becomes INS navigation. At this time, the inertial navigation enters the working state and can be used.





#### Protocol

After the system is powered on, the normal operating time is 20Hz frequency, RS422 communication 115200bps, n, 8, 1 broadcast broadcast navigation data, 48 bytes per frame data as shown in the following table:

Byte number	Content	Data type	Byte count
1-2	Frameheader: 0X5A0XA5	Byte	2
3	Workingstatus:0-Monitoring		
	status,1-Staticalignment,2-INS	Byte	1
	navigation		
4-5	Inertialpitchangle([-90,90]	-	2
	degrees) in units of 0.01 degrees		Z
	Inertialguiderollangle		
6-7	([-180,180]degrees), unit 0.01		2
	degree		
	Inertialazimuth([-180,180]		
8-9	degrees), northto eastis -, north		2
	to west is +, unitis 0.01 degrees		
10-11	nertialX-axisangularvelocity		
	Wx(degree/s)in units of 0.01		2
	deg/s		
12-13	nertial Y-axis angular velocity		
	Wy(degree/s) in units of 0.01		2
	deg/s		
	nertial Z-axis angular velocity	Short int (low byte first, high byte	
14-15	Wx(degree/s)in units of 0.01		2
	deg/s	last)	
16-17	Inertial X-axis acceleration Ax(m/		2
10-17	s / s), unit0.01m / s/ s		2
18-19	Inertial Y-axisaccelerationAy (m /		2
18-19	s/s), unit 0.01m/s/s		۷
20-21	Inertial Z-axis acceleration Az (m /		2
	s / s), unit0.01m / s/ s		۲
22-23	Speed after GPS satellite		2
	positioning, unit0.1m/sec		2
24-25	GPS trackangle([-180,180]		
	degrees), north to the east is -,		2
	north to west is +, the unit is 0.1		2
	degrees		
26-27	GPS satellite number		2
28-29	GPS altitude, unit 0.1 m		2
		Float(low byte first, high byte after), value range [-90,90]	
30-33	Latitude after GPS satellite		4
	positioning, unit degree	degrees, the north latitude is +	4
		and the south latitude is	
	Longitudo offer CDS estallita	Float (low byte first, high byte	
34-37	Longitude after GPS satellite	after), value range[-180,180]	4
	positioning, unit degree	degrees, east longitude is +, west	

#### **Executive standard**

- Enterprise Quality System Standard: ISO9001:2008 Standard(Certificate No.:10114Q16846ROS)
- CE certification (certificate number: 3854210814)
- ROHS (certificate number: SO81426003)
- GJB 2426A-2004 Fiber Optic Gyro Test Method
- GBT 18459-2001 sensor main static performance index calculation method
- JF 1059-1999 Evaluation and Expression of Measurement Uncertainty
- GBT 14412-2005 mechanical vibration and shock mechanical installation of accelerometer
- General requirements for GJB 450A-2004 equipment reliability
- Quality control of key parts and important parts of GJB 909A
- GJB 899 Reliability Qualification and Acceptance Test
- GJB 150-3A high temperature test
- GJB 150-4A low temperature test
- GJB 150-8A rain test
- GJB 150-12A dust test
- GJB 150-16A vibration test
- GJB 150-18A impact test
- GJB 150-23A Tilt and Swing Test
- GB/T 17626-3A RF electromagnetic radiation immunity test
- GB/T 17626-5A surge (hit) impulse immunity test
- GB/T 17626-8A power frequency magnetic field immunity test
- GB/T 17626-11A voltage dips, short interruptions and voltage changes immunity
- GB/T 2423.22-2012 Environmental testing Part 2:test method Test N:temperature change (IEC 60068-2-14:2009,IDT)
- GB/T 10125-2012Artificial atmosphere corrosion test Salt spray test (ISO 9227:2006,IDT)

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